

Diameter of Inlet $d_1 = 300 \text{ mm} = 0.3 \text{ m}$

$$A_1 = \frac{\pi d^2}{4} = \frac{3.142 \times 0.3^2}{4} = 0.0707 \text{ m}^2$$

Diameter of Throat $d_2 = 150 \text{ mm} = 0.15 \text{ m}$

$$A_2 = \frac{\pi d^2}{4} = \frac{3.142 \times 0.15^2}{4} = 0.01767 \text{ m}^2$$

Stn
No. 1
Area = 13.6

$$S.A = 0.9$$

Reading of differential manometer, $y = 250 \text{ mm} = 0.25 \text{ m}$.
The differential "h" is given by;

$$h = \left(\frac{P_1}{\rho} + z_1 \right) - \left(\frac{P_2}{\rho} + z_2 \right)$$

$$= y \left[\frac{\rho_m}{\rho} - 1 \right]$$

$$= 0.25 \left[\frac{13.6}{0.9} - 1 \right]$$

= 3.53 m of oil

A) Discharge of oil Q

$$Q = C_d \times A_1 A_2 \times \sqrt{2gh}$$

$$Q = \frac{C_d \times A_1 A_2 \times \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$
$$Q = \frac{0.98 \times 0.07 \times 0.01767 \times \sqrt{2 \times 9.81 \times 3.53}}{\sqrt{(0.07)^2 - (0.01767)^2}}$$

$$Q = \frac{0.010109}{0.06773}$$

$$Q = 0.149 \text{ m}^3/\text{s}$$

B) Pressure difference between entrance and throat section P1-P2

$$h = \left(\frac{P_1}{\rho} + z_1 \right) - \left(\frac{P_2}{\rho} + z_2 \right) = 3.53$$

$$\therefore h = \left(\frac{P_1}{\rho} - \frac{P_2}{\rho} \right) - (z_1 - z_2) = 3.53$$

$$z_2 - z_1 = 300 \text{ mm} = 0.3 \text{ m}$$

$$\left(\frac{P_1 - P_2}{\rho} \right) - 0.3 = 3.53$$

CIVIL ENGINEERING

- 1) $d_1 = 150 \text{ mm} = 0.15 \text{ m}$, $d_2 = 75 \text{ mm} = 0.075 \text{ m}$, ρ of gravity = 0.8
 $Z_2 - Z_1 = 150 \text{ mm} = 0.15 \text{ m}$, $Q = 40 \text{ Lt/sec} = 0.04 \text{ m}^3/\text{s}$, $C_d = 0.96$
 Pressure difference $P_1 - P_2 = ?$

$$A_1 = \frac{\pi d^2}{4} = \frac{3.142 \times 0.15^2}{4} = 0.01767 \text{ m}^2$$

$$A_2 = \frac{\pi d^2}{4} = \frac{3.142 \times 0.075^2}{4} = 0.00442 \text{ m}^2$$

$$Q = \frac{C_d \times A_1 A_2 \times \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$

$$0.04 = \frac{0.96 \times 0.01767 \times 0.00442 \times \sqrt{2 \times 9.81 \times h}}{\sqrt{(0.01767)^2 - (0.00442)^2}}$$

$$0.04 = 0.96 \times 0.004565 \times 4.429 \sqrt{h}$$

$$h = \frac{(0.04)^2}{(0.96 \times 0.004565 \times 4.429)^2}$$

$$h = 4.247 \text{ m}$$

$$h = \left(\frac{P_1}{w} + Z_1 \right) - \left(\frac{P_2}{w} + Z_2 \right)$$

$$h = \left(\frac{P_1}{w} + \frac{P_2}{w} \right) + (Z_1 - Z_2)$$

$$4.247 = \left(\frac{P_1 - P_2}{\rho g} \right) - 0.15$$

$$4.247 + 0.15 = \frac{P_1 - P_2}{\rho g}$$

$$4.397 = \frac{P_1 - P_2}{\rho g}$$

$$P_1 - P_2 = 4.397 \times (0.8 \times 1000 \times 9.81)$$

$$P_1 - P_2 = 4.397 \times 7848$$

$$P_1 - P_2 = 34.5 \text{ (KN/m}^2\text{)} \\ 345 \text{ N/m}^2$$

$$\frac{P_1 - P_2}{W} = 3.53 + 0.3$$

$$\frac{P_1 - P_2}{W} = 3.83$$

$$P_1 - P_2 = 3.83 W$$

$$P_1 - P_2 = 3.82 \times 9.81 \times 0.9$$

$$= 33.8 \text{ kN/m}^2$$

$$P_1 - P_2 = 33.8$$

$$(z + \beta) - (z + \beta)$$

$$\left[\frac{z + \beta}{z} - 1 \right]$$

$$\left[\frac{z + \beta}{z} - 1 \right]$$

dir to air

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