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DEPARTMENT: MECHANICAL ENGINEERING

COURSE: FLUID MECHANICS

ASSIGNMENT NO 2

① Relative density = 0.8

Inlet diameter $d_1 = 150\text{mm} = 150 \times 10^{-3}\text{m}$

Throat diameter $d_2 = 75\text{mm} = 75 \times 10^{-3}\text{m}$

Actual flowrate $Q_{act} = 40\text{L/sec} = 0.04\text{m}^3/\text{sec}$

$C_d = 0.96$

$z_2 - z_1 = 150\text{mm} = 0.15\text{m}$

$$A_1 = \frac{\pi d_1^2}{4} = \frac{\pi \times (150 \times 10^{-3})^2}{4}$$

$$= 0.0177\text{m}^2$$

$$A_2 = \frac{\pi d_2^2}{4} = \frac{\pi \times (75 \times 10^{-3})^2}{4}$$

$$= 4.419 \times 10^{-3}\text{m}^2$$

$$Q = \frac{C_d A_1 A_2 \sqrt{2gh}}{\sqrt{(A_1^2 - A_2^2)}}$$

$$\therefore Q \sqrt{(A_1^2 - A_2^2)} = C_d A_1 A_2 \sqrt{2gh}$$

$$\sqrt{2gh} = \frac{Q \sqrt{(A_1^2 - A_2^2)}}{C_d A_1 A_2}$$

$$2gh = \left(\frac{Q \sqrt{(A_1^2 - A_2^2)}}{C_d A_1 A_2} \right)^2$$

$$h = \left(\frac{Q \sqrt{(A_1^2 - A_2^2)}}{C_d A_1 A_2} \right)^2 \frac{1}{2g}$$

$$h = \left(\frac{0.04 \sqrt{(0.0177)^2 - (4.419 \times 10^{-3})^2}}{0.96 \times 0.0177 \times 4.419 \times 10^{-3}} \right)^2 \frac{1}{2 \times 9.81}$$

$$h = \left(\frac{6.855799 \times 10^{-4}}{7.50876 \times 10^{-5}} \right)^2 \frac{1}{2 \times 9.81}$$

$$2 \times 9.81$$

$$\frac{(9.130)^2}{2 \times 9.81} = 83.3569$$

$$19.62$$

$$= 4.24 \text{ m}$$

Then,

$$h = \left(\frac{P_1}{K_1} + Z_1 \right) - \left(\frac{P_2}{K_2} + Z_2 \right)$$

$$h = \left(\frac{P_1}{K_1} - \frac{P_2}{K_2} \right) + (Z_1 - Z_2)$$

$$4.24 = \frac{P_1 - P_2}{K_1} + (Z_1 - Z_2)$$

$$\frac{P_1 - P_2}{K_1} = 4.24 + (Z_2 - Z_1)$$

$$\frac{P_1 - P_2}{K_1} = (4.24 + 0.15) \times 10.0 =$$

$$\frac{P_1 - P_2}{K_1} = 4.39$$

$$P_1 - P_2 = 4.39 \times K_1$$

$$P_1 - P_2 = 4.39 \times (0.8 \times 9.81 \times 1000)$$

$$P_1 - P_2 = 4.39 \times 7848$$

$$P_1 - P_2 = 34452.72 \text{ N/m}^2$$

② Inlet d ; $d_1 = 300 \text{ mm} = 300 \times 10^{-3} \text{ m}$

Throat diameter $d_2 = 150 \text{ mm} = 150 \times 10^{-3} \text{ m}$

$$A_1 = 0.07069 \text{ m}^2$$

$$A_2 = 0.0177 \text{ m}^2$$

$$C_d = 0.98$$

S.G. of mercury = 13.6

S.P. of oil = 0.9

Differential manometer = 250 mm = 0.25 m

$$h \cdot \left[\frac{\text{S.G.} - 1}{\text{S.G.}} \right]$$

$$h = \left[\frac{13.6}{0.8} - 1 \right] y$$

$$h = (14.11) \times 0.25$$

$$h = 3.528 \text{ m}$$

(i) The discharge of oil =

$$Q = \frac{C_d A_1 A_2 \sqrt{2gh}}{\sqrt{(A_1^2 - A_2^2)}}$$

$$Q = \frac{0.98 \times 0.07069 \times 0.0177 \times \sqrt{2 \times 9.81 \times 3.528}}{\sqrt{(0.07069^2 - (0.0177)^2)}}$$

$$Q = \frac{0.0102}{0.0689}$$

$$Q = 0.149 \text{ m}^3/\text{s}$$

(ii) The pressure difference between the entrance section and the throat section.

$$h = \left(\frac{P_1}{\rho} + Z_1 \right) - \left(\frac{P_2}{\rho} + Z_2 \right)$$

$$3.528 = \left(\frac{P_1}{\rho} - \frac{P_2}{\rho} \right) - (Z_1 - Z_2)$$

$$3.528 + (Z_2 - Z_1) = \left(\frac{P_1}{\rho} - \frac{P_2}{\rho} \right)$$

Recall, $Z_2 - Z_1 = 300 \text{ mm} = 0.3 \text{ m}$

$$(3.528 + 0.3) = \left(\frac{P_1 - P_2}{\rho} \right)$$

$$P_1 - P_2 = 3.828 \rho$$

$$\rho = 9810 \text{ kg/m}^3$$

$$P_1 - P_2 = 3.828 \times 9810$$

Recall, $\rho = 9810 \text{ kg/m}^3$

$$P_1 - P_2 = 3.828 \times 9810 \times 0.9$$

$$P_1 - P_2 = 33.79 \text{ kN/m}^2$$