

Alcde 016501 Olgoiwa 18/eng02/020
COMPUTER ENGINEERING

1 GIVEN SP OF gravity 0.8, $D_1 = 150\text{mm}$, $D_2 = 75\text{mm}$, $Z_1 = 0.15\text{m}$

$Z_2 = 2.15\text{m}$, $Q = 40\text{lit/sec} = 0.04\text{m}^3/\text{s}$, $C_d = 0.96$

Pressure difference ($P_1 - P_2$)

$$A_1 = \frac{\pi D_1^2}{4} = \frac{\pi \times 0.15^2}{4} = 0.01767\text{m}^2$$

$$A_2 = \frac{\pi D_2^2}{4} = \frac{\pi \times 0.075^2}{4} = 0.00442\text{m}^2$$

$$Q = C_d \times A_1 - A_2 \times \sqrt{2gh}$$

$$0.04 = 0.96 \times 0.01767 \times \sqrt{2 \times 9.81 \times h}$$

$$\sqrt{0.01767^2 - 0.00442^2}$$

$$0.04 = 0.96 \times 0.004565 \times \sqrt{4.429\sqrt{h}}$$

$$h = \left(\frac{0.04}{0.004565 \times 0.96} \right)^2 = 4.247\text{m}$$

$$h = \left(\frac{P_1}{\rho g} + z_1 \right) - \left(\frac{P_2}{\rho g} + z_2 \right) \quad h = \left(\frac{P_1}{\rho g} - \frac{P_2}{\rho g} \right) + (z_1 - z_2)$$

$$4.247 = \left(\frac{P_1 - P_2}{\rho g} \right) - 0.15$$

$$4.247 + 0.15 = \frac{P_1 - P_2}{\rho g} \quad (4.247 + 0.15) \rho g = P_1 - P_2$$

$$P_1 - P_2 = (0.8 \times 1000 \times 9.81) (4.247 + 0.15)$$

$$P_1 - P_2 = 34.51\text{KN/m}^2$$

2) DIAMETER OF INLET $D_1 = 300\text{mm} = 0.3\text{m}$

$$\text{Area of inlet } A_1 = \frac{\pi \times D_1^2}{4} = \frac{\pi \times 0.3^2}{4} = 0.0707\text{m}^2$$

$$\text{Diameter of throat } D_2 = 150\text{mm} = 0.15\text{m}$$

$$\text{Area of inlet } A_2 = \frac{\pi D_2^2}{4} = \frac{\pi \times 0.15^2}{4} = 0.01767\text{m}^2$$

Specific gravity of heavy liquid (mercury) in II

Tube manometer $S_L = 13.6$

Specific gravity of liquid (oil) flowing through pipe $S_F = 0.9$

Reading of differential manometer, $y = 250\text{mm} = 0.25\text{m}$

The differential 'h' is given by;

$$h = \left(\frac{P_1}{\rho g} + z_1 \right) - \left(\frac{P_2}{\rho g} + z_2 \right)$$

$$= y \left[\frac{S_L}{S_F} - 1 \right] = 0.25 \left[\frac{13.6}{0.9} - 1 \right] = 3.53 \text{ OF OIL}$$

a) Discharge of oil Q

$$\text{Using the relation: } Q = \frac{C_d \times A_1 \times A_2 \times \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$

$$Q = \frac{0.98 \times 0.07 \times 0.01767 \times \sqrt{2 \times 9.81 \times 3.53}}{\sqrt{0.07^2 - 0.01767^2}}$$

$$Q = 0.1489\text{m}^3/\text{s}$$

b) Pressure difference between entrance and throat

Section $P_1 - P_2$ we all know that

$$h = \left(\frac{P_1}{\rho g} + z_1 \right) - \left(\frac{P_2}{\rho g} + z_2 \right) = 3.53$$

$$h = \left(\frac{P_1}{\rho g} - \frac{P_2}{\rho g} \right) - (z_2 - z_1) = 3.53$$

$$z_2 - z_1 = 300\text{mm} = 0.3\text{m}$$

$$\left(\frac{P_1 - P_2}{\rho g} \right) - 0.3 = 3.53 = \frac{P_1 - P_2}{\rho g} = 3.53 + 0.3$$

$$\frac{P_1 - P_2}{\rho g} = 3.83$$

$$P_1 - P_2 = 3.83 \rho g$$

$$P_1 - P_2 = 3.83 \times 9.81 \times 0.9$$

$$= 33.81\text{KN/m}^2$$